

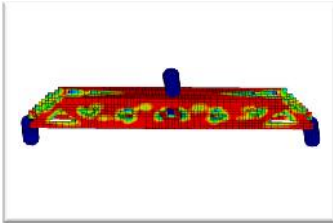
Application of the equivalent static load method for impact problems with GENESIS and LS-DYNA

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DYNAmore GmbH

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MAGNA STEYR Engineering AG & Co KG

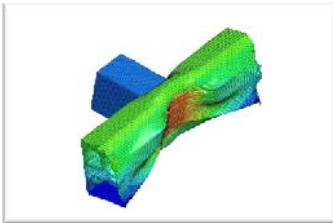
Hanau, 11th April 2013
Automotive CAE Grand Challenge

Outline



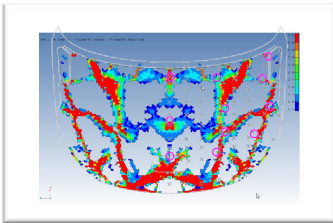
Introduction

Equivalent Static Load Method



Case Study 1

Extrusion Profile Optimization, Research Project Crash-Topo



Case Study 2

Optimization of an Engine Hood



Summary

Conclusions, Lessons Learned

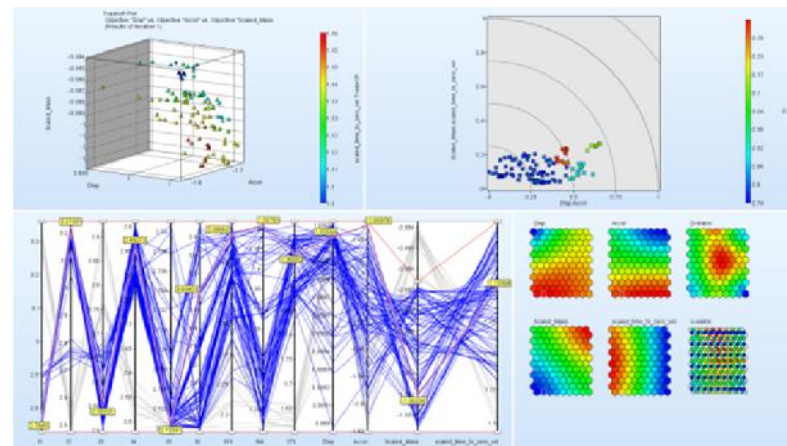
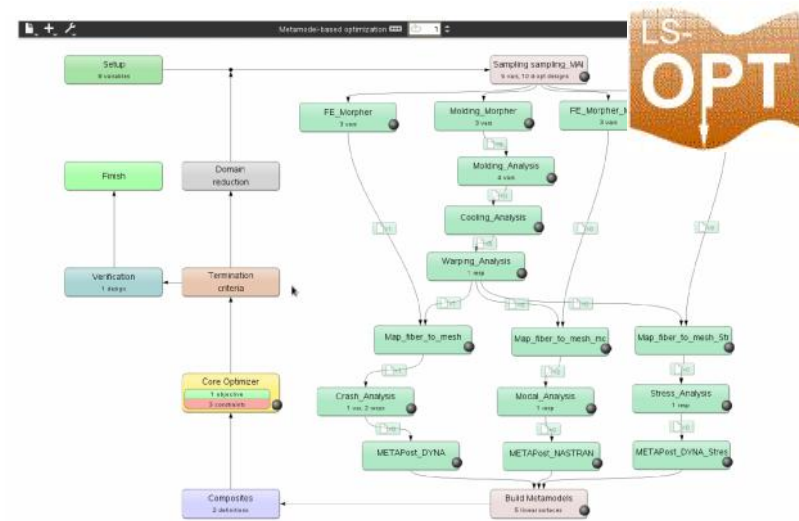
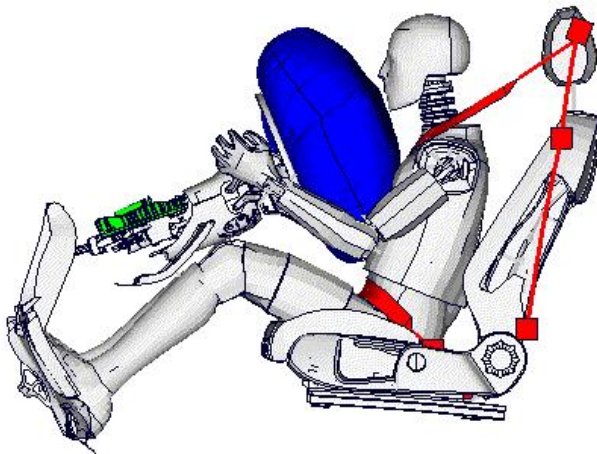
Introduction - Classification of Optimization

Nonlinear (Crash) Optimization

Software Product: LS-OPT

Non-linear / Parametric

- Parameterization of input files
- Shape/Sizing Optimization
- Limited to moderate number of variables (~<50)
- Possible for general nonlinear applications: Crash, Fluid Dynamics, Nonlinear Static/Dynamic



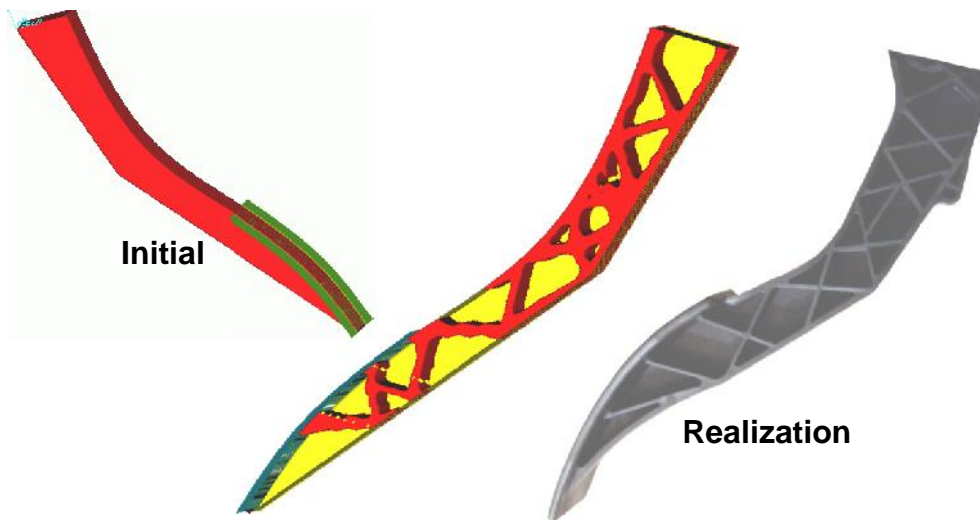
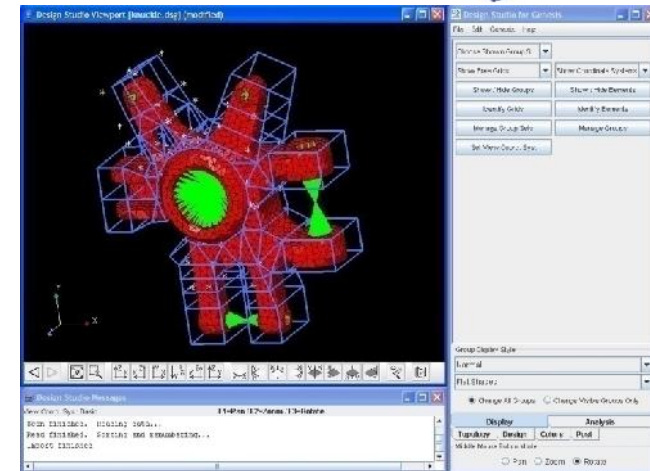
Introduction - Classification of Optimization

Linear Optimization

Software Product: **GENESIS**

Non-Parametric

- Topology / Topometry Optimization
- Usually Linear FE-Problems
- Gradient based solvers – many design variables > 1000000
- CAE-Applications: Static Loads, Frequency Analysis, NVH,...



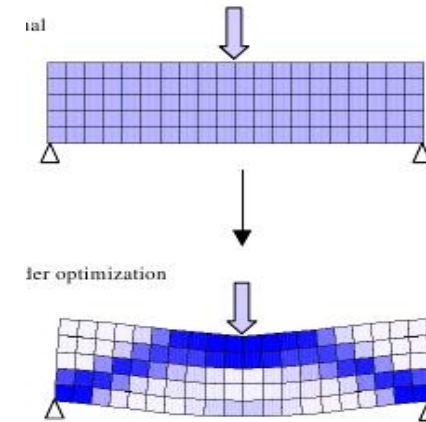
Introduction

Topology/Topometry Optimization for Crash?

For topology optimization each element is a design variable - can be switched on/off

→ many variables

- *Can not be solved with LS-OPT (too many variables)*
- *Can not be solved for crash with gradient based topology solvers such as Genesis (strong non-linearities)*



Two considerable approaches

Equivalent Static Loads Method - ESLM

Hybrid Cellular Automata (HCA)

Product: LS-TASC



Introduction ESL

Idea of the Equivalent Static Load Method

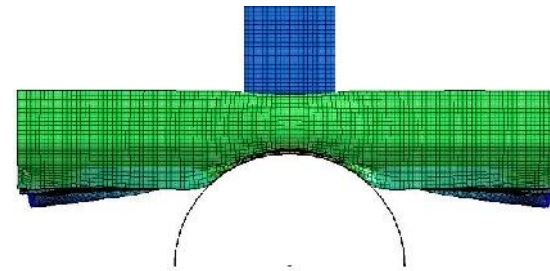
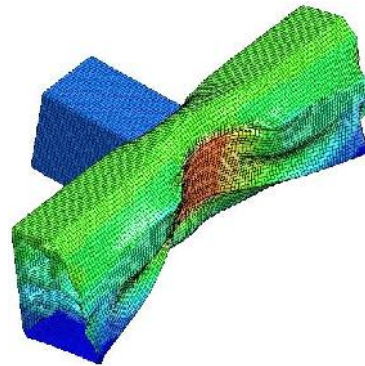
Decomposition of the nonlinear, dynamic optimization problem in

Nonlinear dynamic analysis displacement field

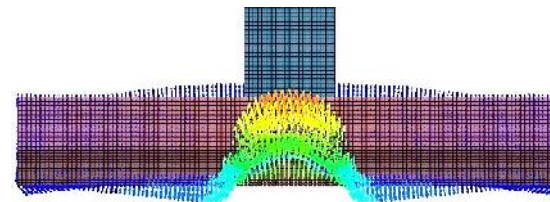
Equivalent static loads for single time steps

„multi load case topology optimization“ with equival. static loads

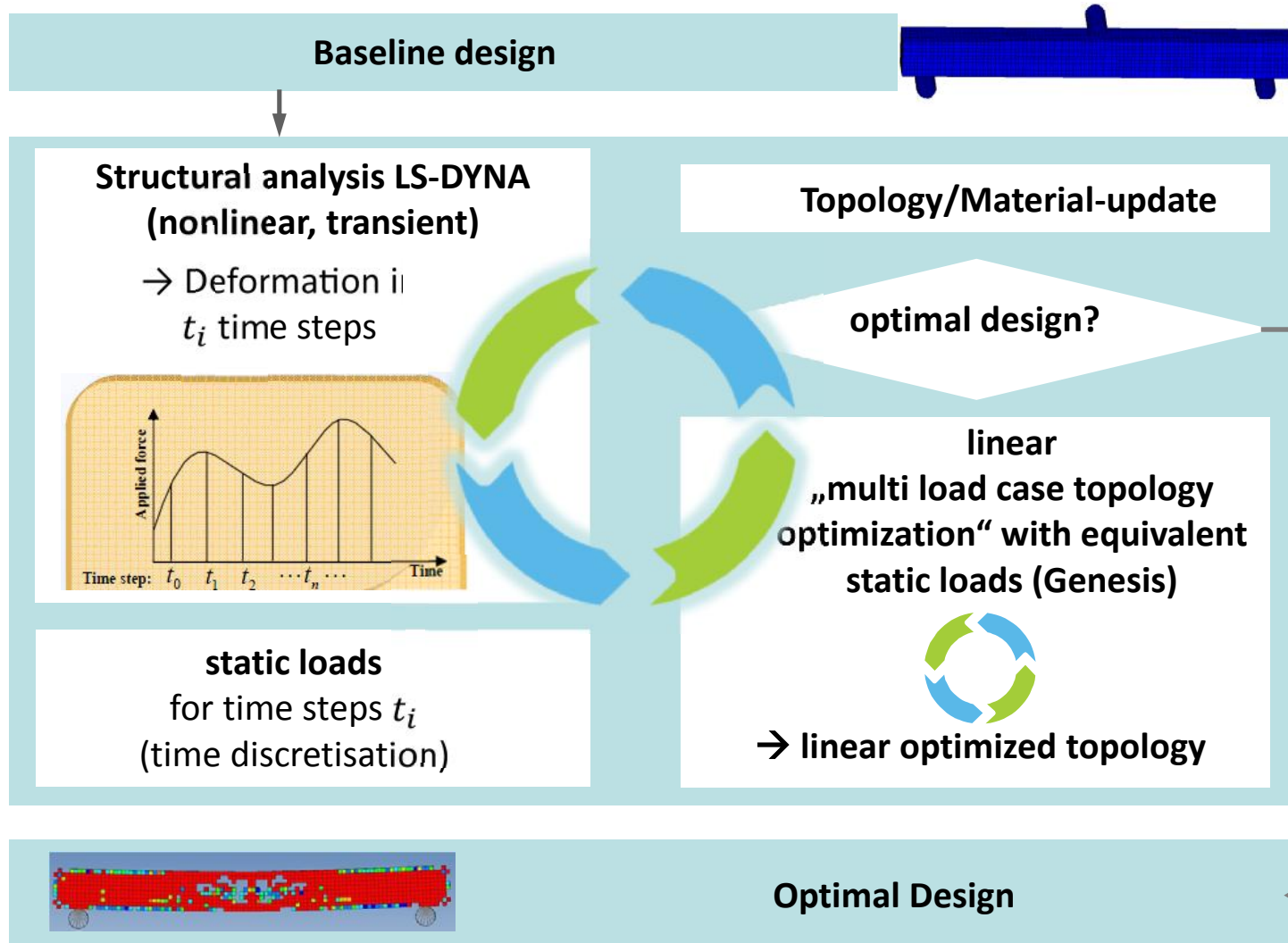
Displacement field:
 $\mathbf{u}_t(\mathbf{x})$



Equivalent static loads:
 $\mathbf{F}_t(\mathbf{x}) = \mathbf{K}_{lin} \mathbf{u}_t(\mathbf{x})$



Introduction ESL

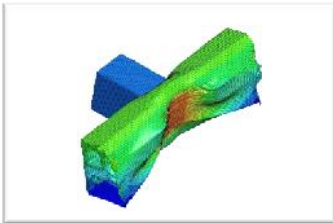


Agenda



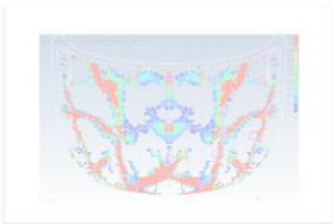
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Equivalent Static Load Method



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Case Study 2

Optimization of an Engine Hood



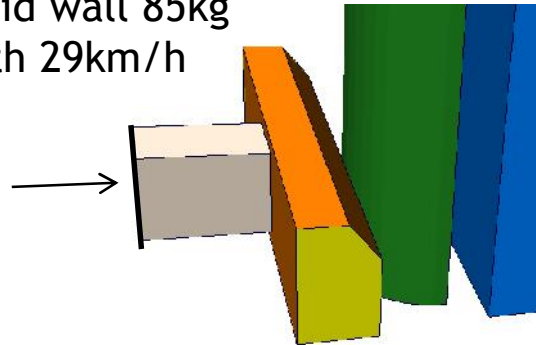
Summary

Conclusions, lessons learned

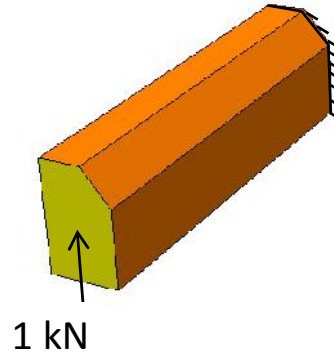
Extrusion Profile Optimization

Load Cases

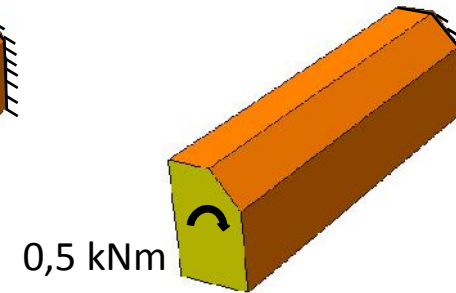
Rigid wall 85kg
with 29km/h



Pole Crash



Bending



Torsion

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Targets

LC Crash: Contact force < 40 kN, time history of contact force as uniform as possible, Intrusion < 70 mm

LC Bending: Displacement < 0.3867 mm

LC Torsion: Wrinkling $< 3.554 \cdot 10^{-3}$ rad

Mass < 2.8 kg

1.6 mm $<$ fillet thickness < 3.5 mm

Extrusion Profile Optimization

Objectives

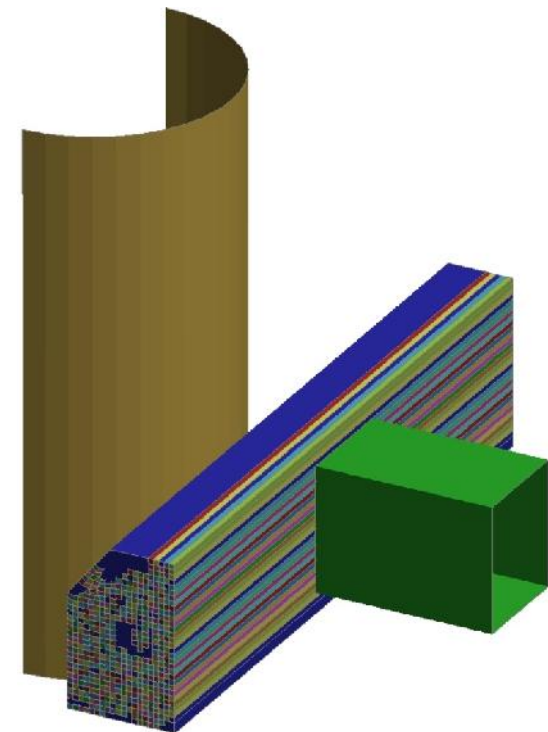
- LC Crash: maximize internal energy
- LC Bending: minimize internal energy
- LC Torsion: minimize internal energy

Constraints

- LC Crash: Intrusion < 70mm
- LC Bending: Displacement < 0.3867mm
- LC Torsion: Wrinkling < $3.554 \cdot 10^{-3}$ rad

Element discretization

- Hexaeder elements with 5mm edge length
- Fully integrated elements



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Extrusion Profile Optimization

Result example with ESL-Method

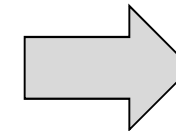
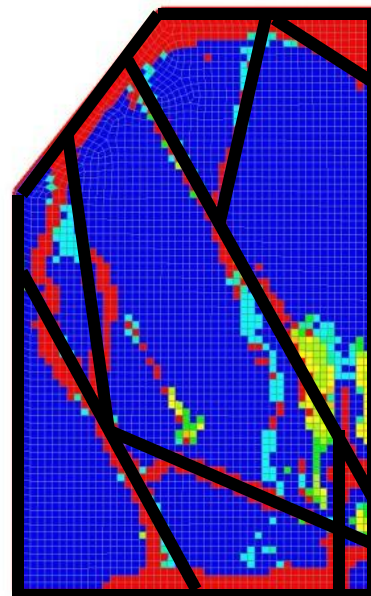
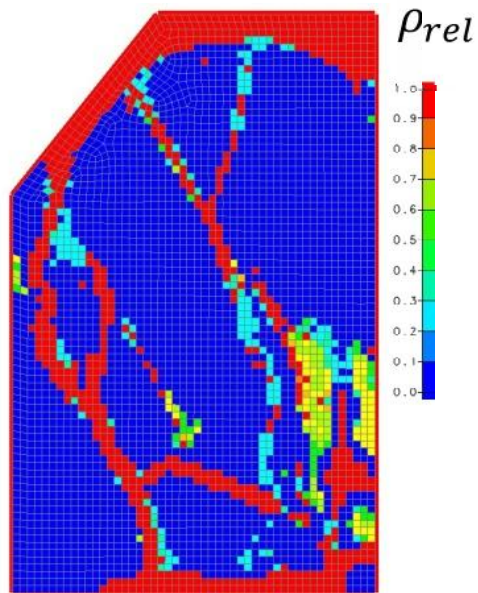
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Optimized relative
density distribution

Possible
interpretation



Results might be transferred to SFE
concept for subsequent shape
optimization with GHT and LS-OPT
- interface has been developed
within research project

Extrusion Profile Optimization

Result example with ESL-Method

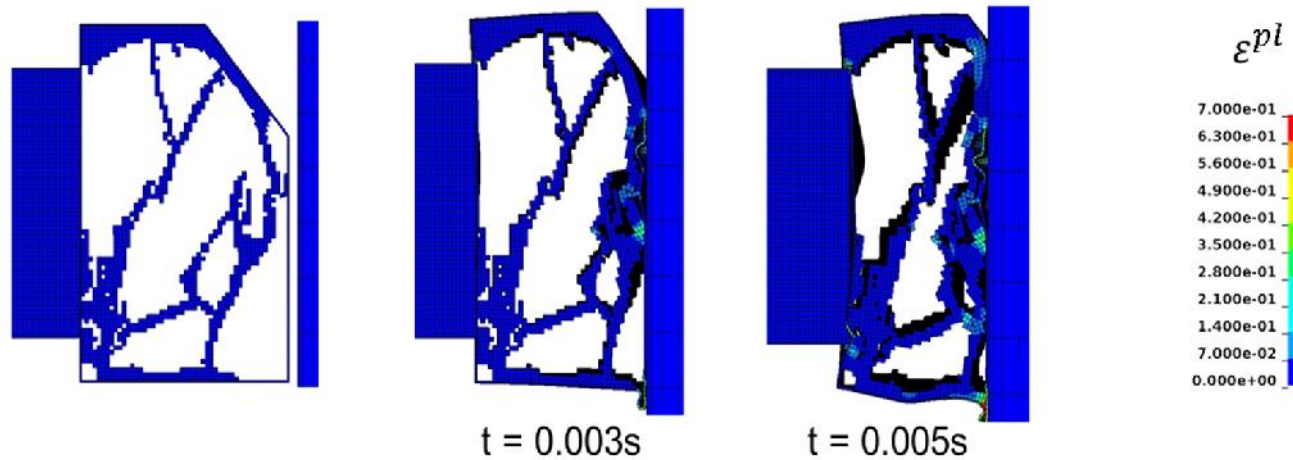
Analysis results of optimized topology

Maximal Intrusion: 67,1 mm (constraint: $d < 70\text{mm}$)

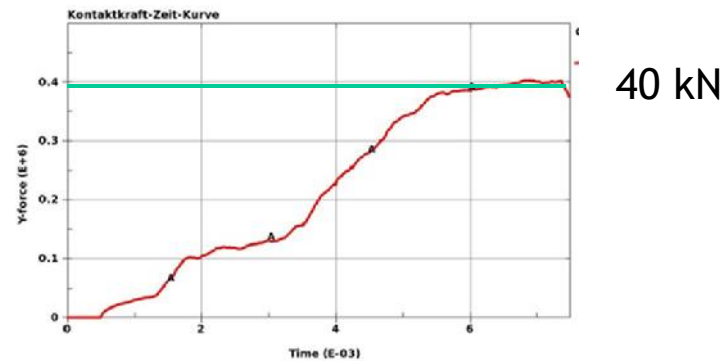
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Maximum contact force: 40,4 kN



Summary

Within the research project „Crash Topo“ topology optimization of extrusion profiles, mainly on the example of automotive rocker sills, was examined

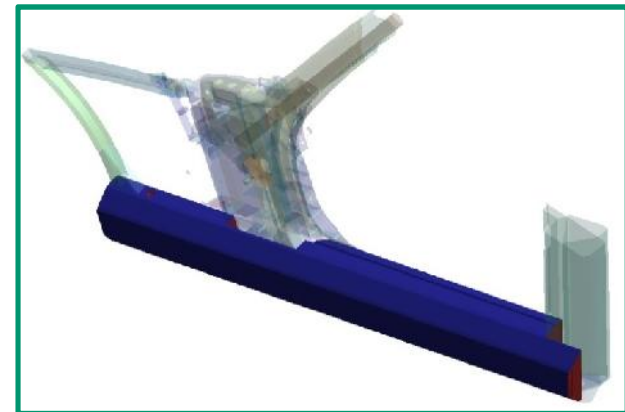


As one new approach for optimization the „Equivalent Static Load Method“ was applied

An automated process with LS-DYNA and Genesis has been setup on an HPC environment

Geometry of rocker sills can be very complex → no straight forward extrusion profiles

Fine resolution (small element size) of solid elements within construction space is required, but lead to many elements (ex.: 1mm el.-length → ~10mio elements)



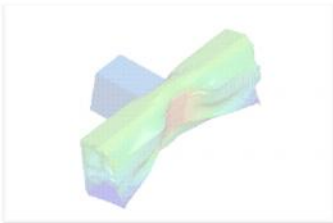
Large buckling of fillets lead to limits of ESL method

Agenda



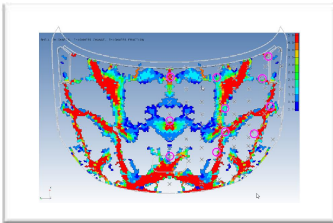
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Equivalent Static Load Method



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Case Study 2

Optimization of an Engine Hood



Summary

Conclusions, lessons learned

Project Information

Joint project between MAGNA STEYR Engineering AG & Co KG and DYNAmore GmbH

Motivation

Development of a standardized method to design an inner hood panel
Method should be able to take into account different package and geometry conditions

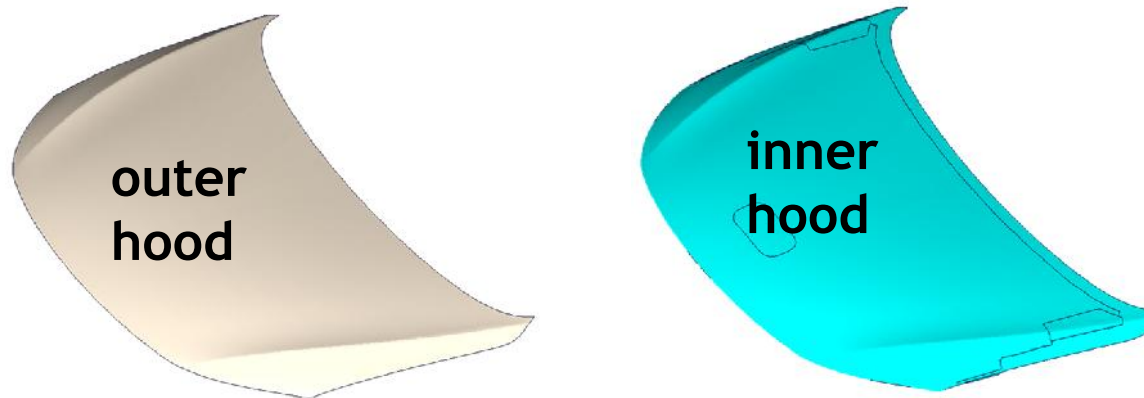
Main load cases are head impact (pedestrian safety) and stiffness

Expected Results

Design of inner hood panel with optimal HIC-value for head impact and stiffness values for static load cases

Outer hood with constant shell thickness $t=0,6\text{mm}$ and material H220

Inner hood is a duplicate of the outer hood with same nodes and coincident elements but separate property with material DX 56D.



Design variables for optimization are thicknesses of every single element (Topometry Optimization).

Variation of thickness between 0,1mm and 5,0mm.

Reduction of number of variables

Clustering of elements \rightarrow 4 neighbouring elements have the same thickness during optimization.

Symmetry constraint in y-direction

LS-DYNA model for nonlinear impact simulation

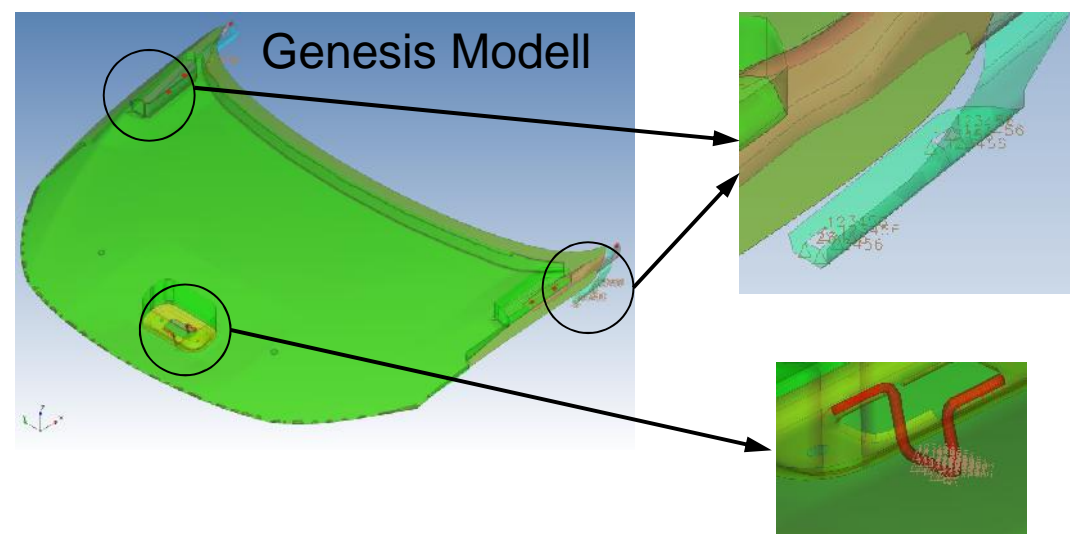
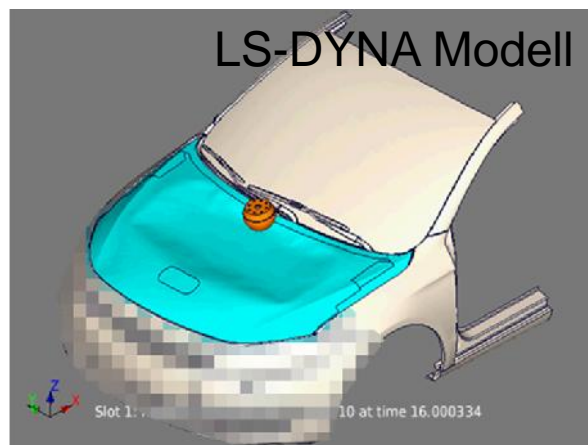
reduced car model with blocking package elements in the engine compartment

Genesis model for optimization with ESL method

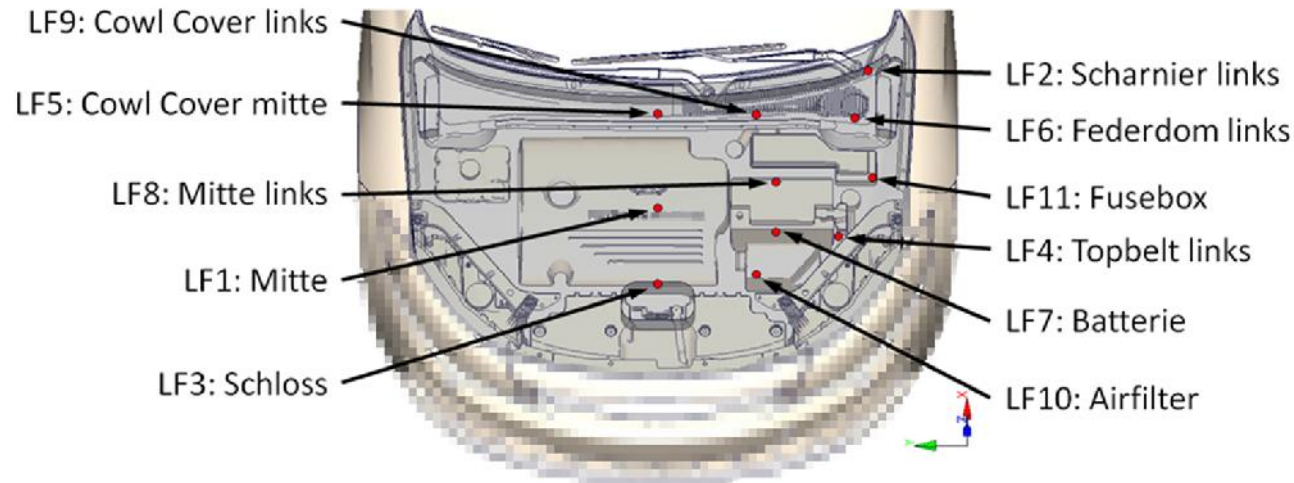
only hood with hinges and lock is considered

support with SPC's on the hinges and the lock

the preceding LS-DYNA simulation has been discretized with 9 equivalent static load cases ($\Delta t=2$ ms)



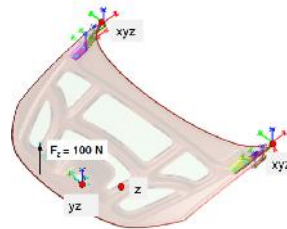
Head impact at 11 points



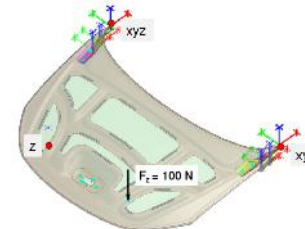
Static loads

- corner bending
- torsion
- bending cross member
- bending longitudinal member

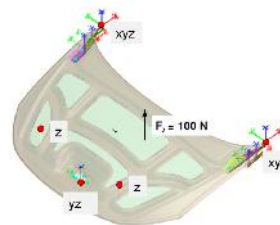
Lastfall: Corner Bending



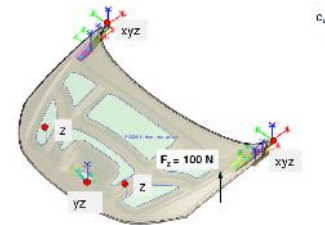
Lastfall: Torsion



Lastfall: Bending Cross Member



Lastfall: Bending Longitudinal Member



HIC-Value can not be used as an objective in linear inner topology optimization loop

Opt. problem formulation for head impact instead

Maximize deformation of the hood by avoiding contact with stiff (rigid) underlying structure

Objective

Maximize strain energy for head impact load cases

Constraints

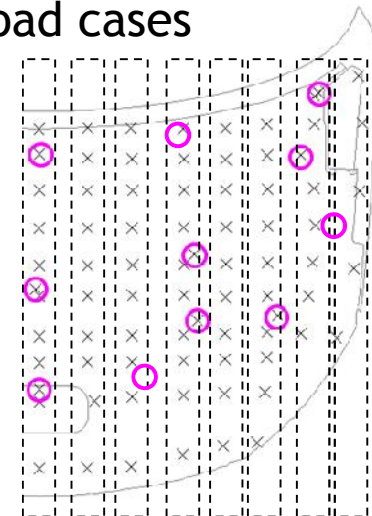
Limits for displacement in z-direction for head impact load cases

About 80 points with maximum feasible deformation

Only for the ESL load cases with large deformation from 6ms on (7 per head impact point)

$11 \text{ (Head impact point)} * 7 \text{ (ESL)} * 80 \text{ (Points with displacement limit)}$
 $= 6160 \text{ (constraints)}$

Limits for displacement of the static load cases



Results

Evaluation of HIC values for each LS-DYNA simulation

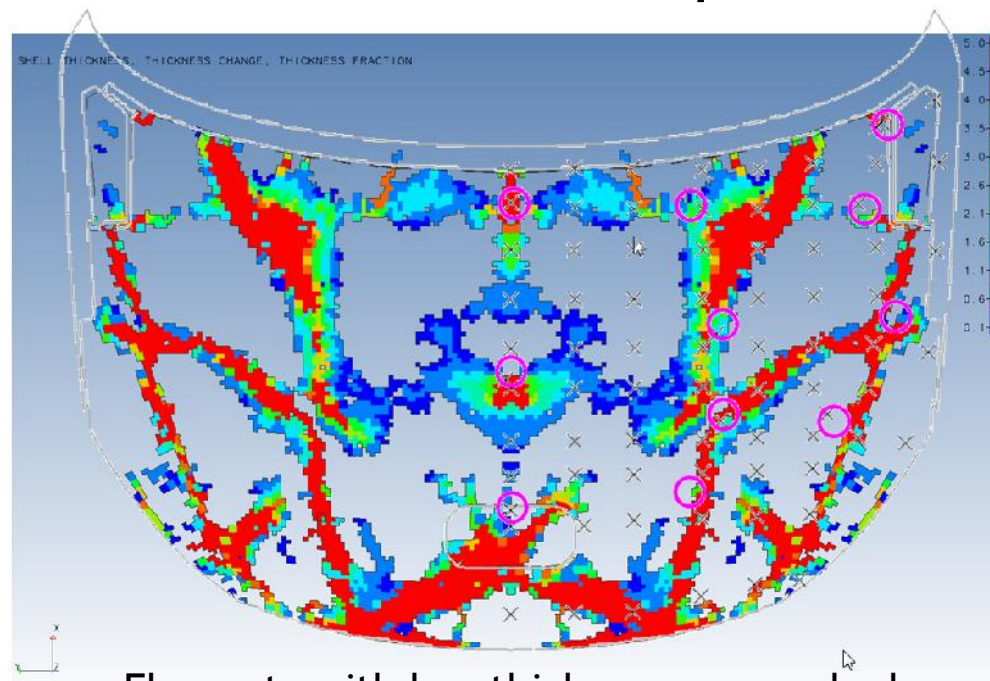
Starting design

Dyna-Rechnung	LF1_Mitte	LF2_Scharnier_li	LF3_Schloss	LF4_Topbelt	LF5_Cowl_Cover	LF6_Federdom	LF7_Batterie	LF8_Mitte_li	LF9_Cowl_li	LF10_Airfilter	LF11_Fusebox	unter 900	900-1000	über 1000	Vmin > 0
0												4	2	3	2

Optimal design

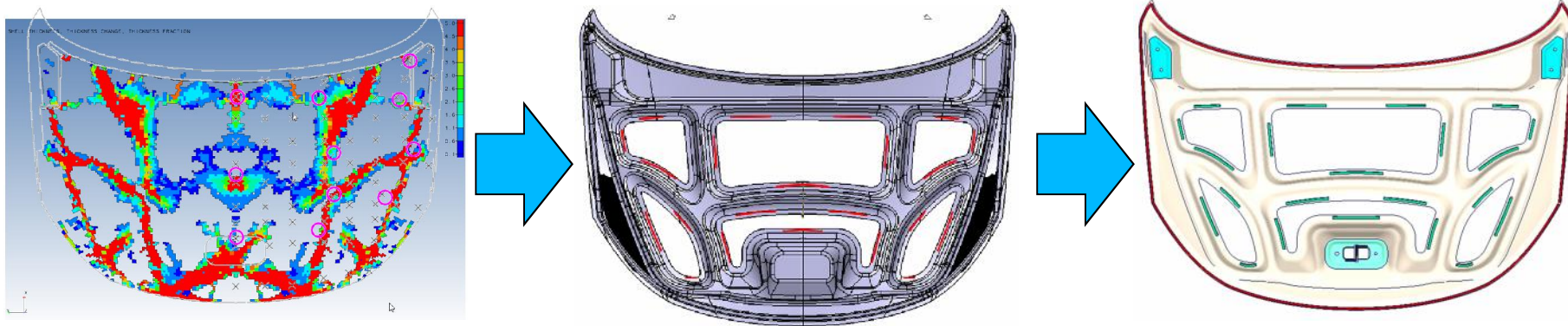
Dyna-Rechnung	LF1_Mitte	LF2_Scharnier_li	LF3_Schloss	LF4_Topbelt	LF5_Cowl_Cover	LF6_Federdom	LF7_Batterie	LF8_Mitte_li	LF9_Cowl_li	LF10_Airfilter	LF11_Fusebox	unter 900	900-1000	über 1000	Vmin > 0
17												8	0	3	0

Element thickness distribution for the optimal solution



Elements with low thickness are masked

Interpretation of CAD-design of the inner hood



LS-DYNA simulation results of the final design

Head impact, HIC values

On average, results of final CAD-design getting a little worse compared to final topometry optimization results

Static loadcases

- torsion → threshold value complied
- corner bending → threshold value complied
- bending cross member → threshold value slightly violated
- bending longitudinal member → threshold value complied

Topometry optimization for the design of the supporting structure of an engine hood has been performed

As a new approach for optimization the „Equivalent Static Load Method“ was applied

An automated process with LS-DYNA for nonlinear pedestrian impact simulations and Genesis for linear topometry optimization was established

The result is a preliminary CAD design of the supporting structure

In a next step nonlinear parameter optimization with LS-OPT will be performed on the basis of the preliminary CAD design to refine functional requirements

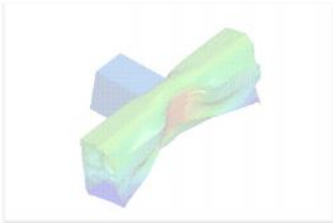
Parameters for the optimization with LS-OPT might be gauge thickness, properties of glue lines, geometric shapes based on morphing, etc.

Agenda



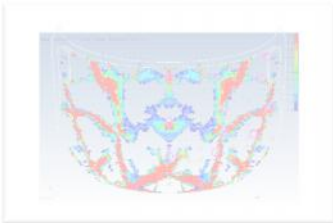
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Equivalent Static Load Method



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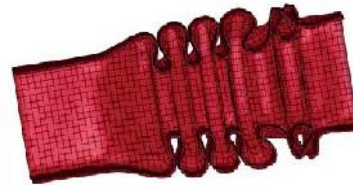
Summary

Conclusions, Lessons Learned

Conclusions

Limit of the ESL-Methodologie

Local buckling/folding where plastic hinges occur leads to out of scale equivalent static loads



Nonlinear Model
(LS-DYNA)

plastic hinge occur after exceeding yield stress

necessary force or moment respectively for large buckling deformation is relatively small

The diagram for the nonlinear model consists of two parts. The first part shows a horizontal line with a central dot representing a plastic hinge. On the left, a vertical double-headed arrow indicates a force applied perpendicular to the line. On the right, a vertical double-headed arrow indicates a force applied perpendicular to the line. The second part shows a V-shaped line with a central dot at the vertex, representing a plastic hinge. Two curved arrows on either side indicate moments applied to the ends of the V-shape.

Linear Model
(Genesis equivalent static loads)

necessary force or moment respectively for same large buckling deformation $\rightarrow \infty$

The diagram for the linear model shows a V-shaped line with a central dot at the vertex, representing a plastic hinge. Two curved arrows on either side indicate moments applied to the ends of the V-shape.

Conclusions

Formulation of Objectives

Objectives are defined for linear optimization. This means, consideration of nonlinear responses are not directly possible

Examples: Minimization of HIC value for head impact is not possible as an objective

Alternative criteria have to be established

Formulation of Constraints

Constraints are defined for linear optimization as well. Consideration of constraints based on nonlinear responses is not possible

Constraints are satisfied for the linear replacement problem. They might be violated for the real nonlinear problem

Automated Model Transition

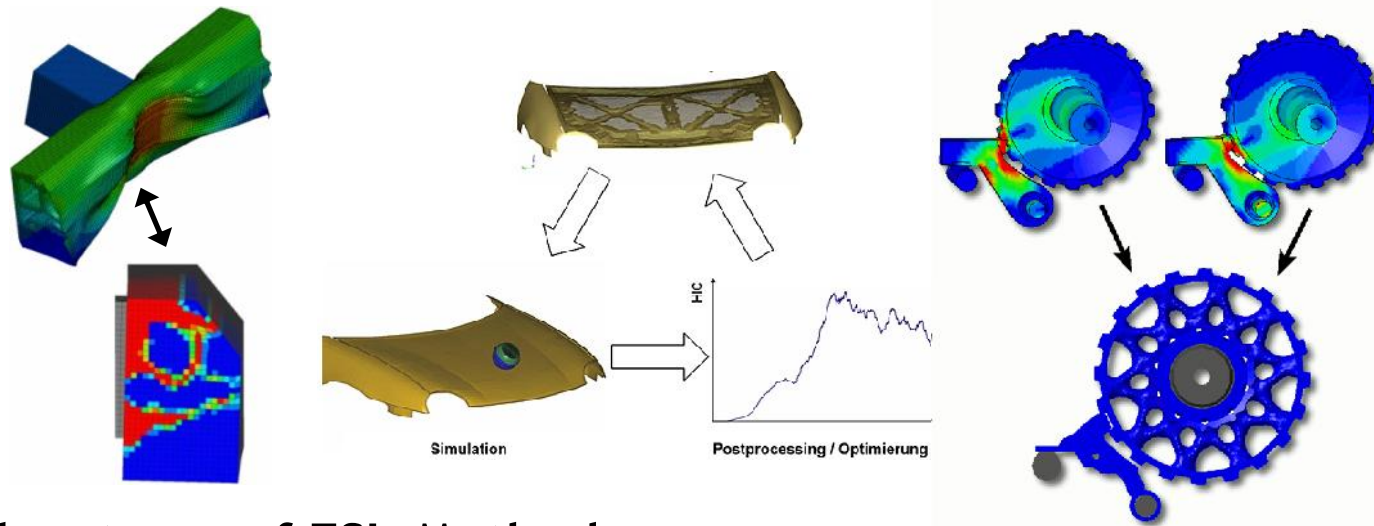
The nonlinear LS-DYNA model has to be translated to a linear Genesis model. Automation of this process is a challenging task. Many Keywords and modelling features of LS-DYNA are supported, but not 100% yet.

Conclusions

ESL-Method is promising

for nonlinear applications with rather moderate deformations or with more extensive buckling, for any contact problems, etc.

Examples: Roof crash test, pedestrian safety load cases, pendulum impact, drop tests, gear wheels ...



Advantages of ESL-Method

Enables Topology/Topometry optimization for nonlinear problems

Size/Shape (parametric) optimization with fewer nonlinear solver calls

Thanks for your attention!

