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An Overview of Ductile Damage Models in LS-DYNA

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overview of ductile damage models in LS-DYNA

implementation of a damage formulation in SAMP

application to a structural part made of thermoplastic material

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ductile damage models in LS-DYNA

isochoric damage models (Lemaitre)

non-isochoric damage models for metals (Gurson)

$$d(\varepsilon_p) \rightarrow v_p = \frac{2 - q_1 f^*}{4 + q_1 f^*}$$

non-isochoric damage models for plastics (SAMP), modeling of crazing

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ductile damage models in LS-DYNA

in ductile damage models the damage variable evolution is defined as some function of the equivalent plastic strain

additional dependencies may exist on the hardening characteristic and/or the stress triaxiality

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Damage Models in LS-DYNA

	DAMAGE EVOLUTION	Path dependent	Non-proportional loading
MAT_81	$d = d(\varepsilon_p)$	yes	no
MAT_81	$d = \int_{\varepsilon_{pd}}^{\varepsilon_p} d_c \frac{d\varepsilon_p}{\varepsilon_{pr} - \varepsilon_{pd}}$	yes	no
MAT_105	$d = \int_{\varepsilon_{pd}}^{\varepsilon_p} \frac{\sigma_{vm}^2 \left(\frac{2}{3}(1+\nu) + 3(1-2\nu) \left(\frac{\sigma_H}{\sigma_{vm}} \right)^2 \right)}{2E(1-d)^2 S} d\varepsilon_p$	yes	yes

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Damage Models in LS-DYNA

	Damage Evolution	Path dependent	Non-proportional loading
MAT_107	$d = \int_0^{\varepsilon_p} \frac{d\varepsilon_p}{\left(d_1 + d_2 e^{d_3 \frac{p}{\sigma_{vm}}} \right)}$	yes	yes
MAT_120	$f = f_0 + \int (1-d) \dot{\varepsilon}_p + \int \frac{f_N}{s_N \sqrt{2\pi}} e^{-\frac{1}{2} \left(\frac{\varepsilon_p - \varepsilon_N}{s_N} \right)^2} \dot{\varepsilon}_p$ $f^* = \begin{cases} f & f \leq f_c \\ f_c + \frac{q_1}{f_F - f_c} (f - f_c) & f > f_c \end{cases}$ $d := q_1 f^* \leq d_c = q_1 f^*(f_F) = 1$	yes	yes

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compatibility condition

$$\left. \begin{aligned} \dot{\varepsilon}_{vp} &= -\dot{\lambda} \frac{\partial f}{\partial p} \\ \dot{\varepsilon}_p &= \dot{\lambda} \frac{\partial f}{\partial \sigma_{vm}} \end{aligned} \right\} \Rightarrow \dot{\varepsilon}_{vp} \frac{\partial f}{\partial \sigma_{vm}} = -\dot{\varepsilon}_p \frac{\partial f}{\partial p}$$

compatibility condition after Taylor series expansion allows to express volumetric plastic strain as a function of effective plastic strain, triaxiality, and void fraction

$$\dot{\varepsilon}_{vp} = -\dot{\varepsilon}_p \frac{\frac{\partial f}{\partial p}}{\frac{\partial f}{\partial \sigma_{vm}}} = -\dot{\varepsilon}_p \frac{9}{4} q_1 q_2^2 f^* \frac{p}{\sigma_{vm}}$$

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Damage Evolution for uniaxial tension

MAT_81	$d = d(\varepsilon_p)$
MAT_81	$d = \int_{\varepsilon_{pd}}^{\varepsilon_p} d_c \frac{d\varepsilon_p}{\varepsilon_{pr} - \varepsilon_{pd}}$
MAT_105	$d = \int_{\varepsilon_{pd}}^{\varepsilon_p} \frac{\sigma_{vm}^2(\varepsilon_p) \left(\frac{2}{3}(1+\nu) + 3(1-2\nu) \left(\frac{1}{3} \right)^2 \right)}{2ES(1-d)^2} d\varepsilon_p$ $d = \int_{\varepsilon_{pd}}^{\varepsilon_p} \frac{\sigma_y^2(\varepsilon_p)}{2ES} d\varepsilon_p$

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Damage Evolution for uniaxial tension

MAT_107	$d = \frac{1}{\left(d_1 + d_2 e^{-d_3 \frac{1}{3}}\right)} \int_0^{\varepsilon_p} d\varepsilon_p$
MAT_120	$f \approx f_0 + \int -(1-f)q_1 f^* q_2^2 \frac{9}{4} \frac{p}{\sigma_{vm}} \dot{\varepsilon}_p + \int \frac{f_N}{s_N \sqrt{2\pi}} e^{-\frac{1}{2} \left(\frac{\varepsilon_p - \varepsilon_N}{s_N}\right)^2} \dot{\varepsilon}_p$ $f^* = \begin{cases} f & f \leq f_c \\ f_c + \frac{q_1}{f_f - f_c} (f - f_c) & f > f_c \end{cases}$ $d = q_1 f^* \leq 1$

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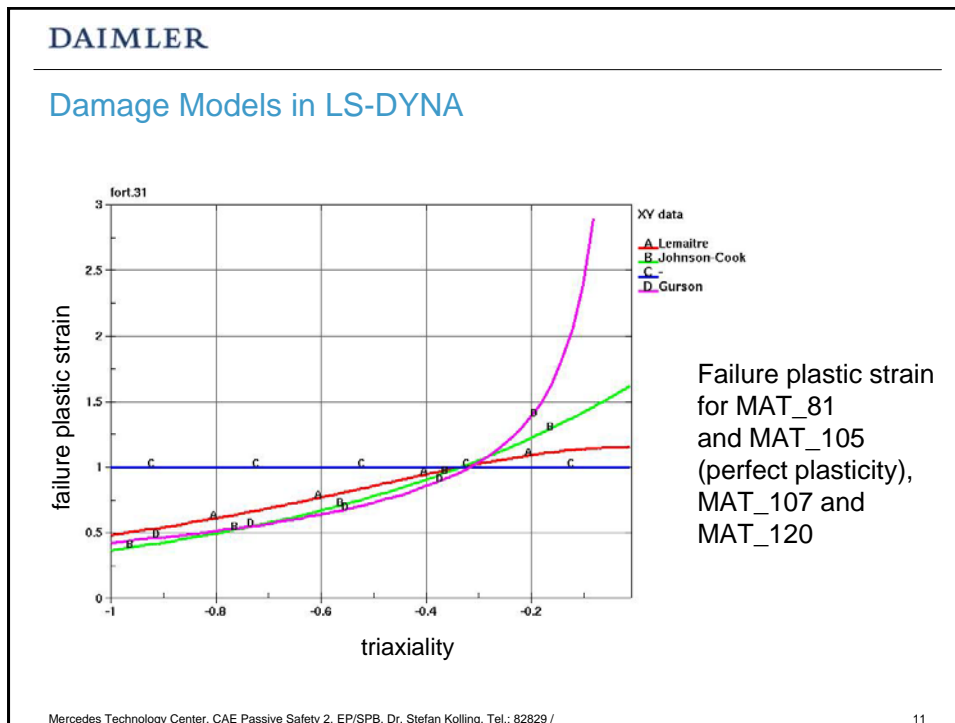
Damage Evolution for uniaxial tension

XY data

- A Lemaitre bilinear law
- B Gurson
- C Lemaitre perfect plasticity

Damage in function of longitudinal plastic strain under uniaxial tension

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SAMP damage model

$$\varepsilon_{pf} = DC(\dot{\varepsilon}_{pt}) LCID_TRI\left(\frac{p}{\sigma_{vm}}\right) LCID_LC(l_c)$$

$$\varepsilon_{pf0} = DC(0)$$

$$LCID_TRI\left(-\frac{1}{3}\right) = 1$$

$$LCID_LC(l_{c0}) = 1$$

User defines the plastic strain at failure as a function of plastic strain rate, stress triaxiality and characteristic element length using up to 3 loadcurves

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SAMP damage model

$$d = LCID_D(\varepsilon_{pt}) \leq 1$$

$$d(\varepsilon_{pt}) = 1 - \frac{E(\varepsilon_{pt})}{E}$$

User defines damage as a function of plastic strain under uniaxial tension, the loadcurve can be determined from tensile experiments with unloading

elements will erode if the damage variable becomes equal to or exceeds the critical value (1 if no failure strain is defined)

$$d(\varepsilon_{pt}) \leq d_c = LCID_D(\varepsilon_{pf0})$$

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SAMP damage model

$$\int \frac{1}{d(\varepsilon_{pf})} \frac{\partial d}{\partial \varepsilon_{pt}} d\varepsilon_{pt} \leq 1$$

$$\left. \begin{array}{l} \frac{p}{\sigma_{vm}} = c^{te} \\ \dot{\varepsilon}_p = c^{te} \\ l_c = c^{te} \end{array} \right\} \Rightarrow d \leq d(\varepsilon_{pf})$$

The damage model in SAMP combines the data from the uniaxial unloading tests and the measurements of failure strain under different loading conditions

for a constant triaxiality and strain rate, the user-defined strain to failure will be reached exactly

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SAMP damage model

SAMP allows to emulate most other damage models by using adequate load curves

in MAT_120 volumetric plastic strain will occur as a consequence of the damage model

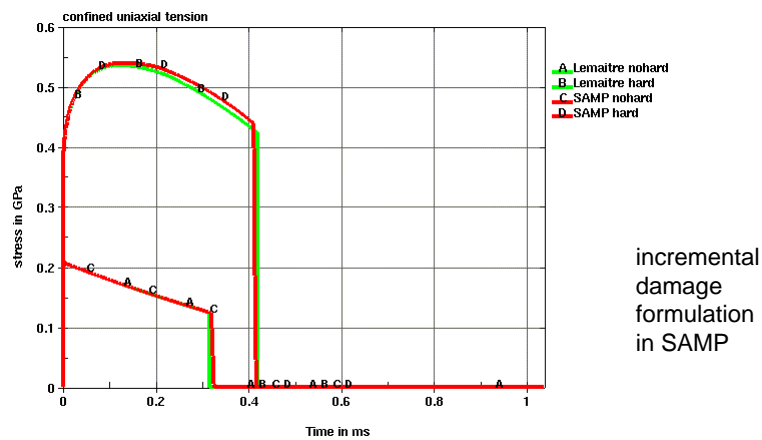
in SAMP the plastic Poisson ratio is defined as an extra load curve which directly defines the plastic flow surface

single element tests illustrate the equivalence of different approaches

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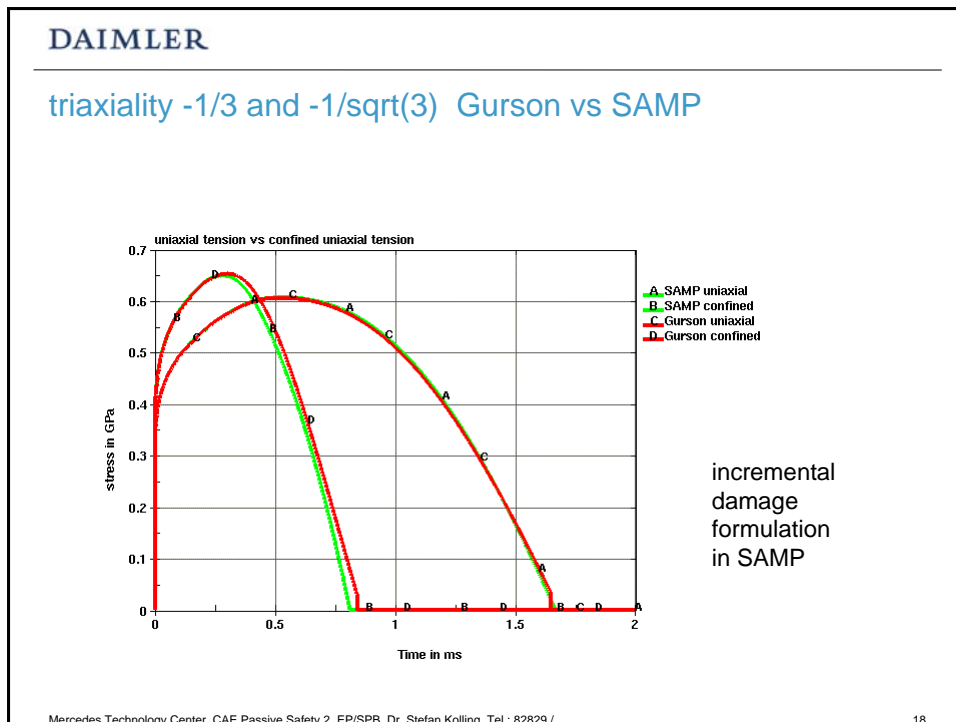
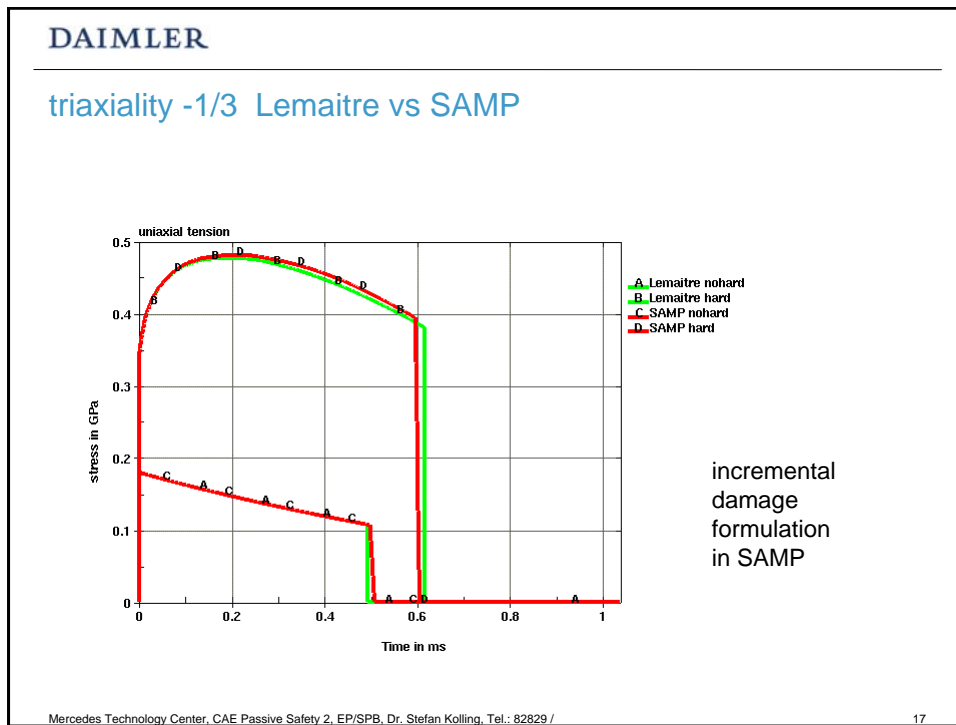
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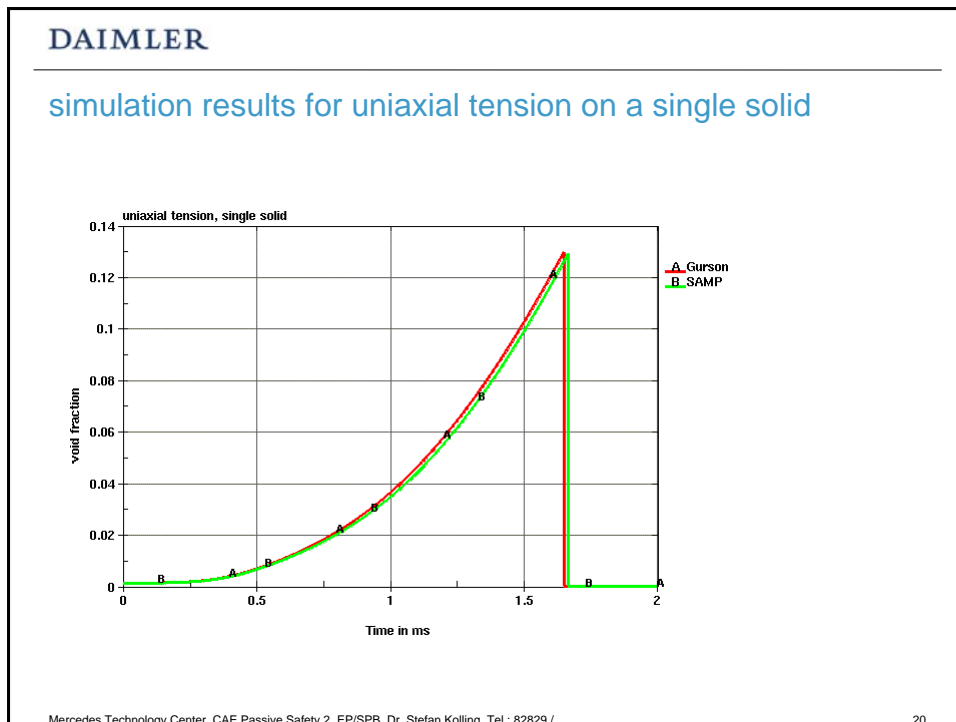
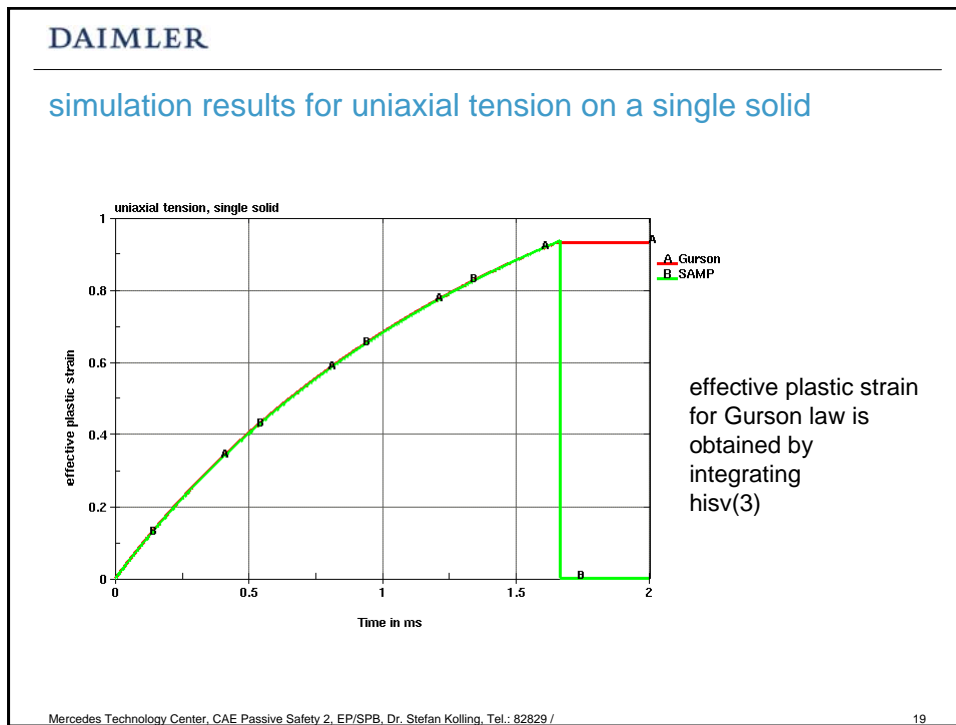
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triaxiality $-1/\sqrt{3}$ Lemaitre vs SAMP

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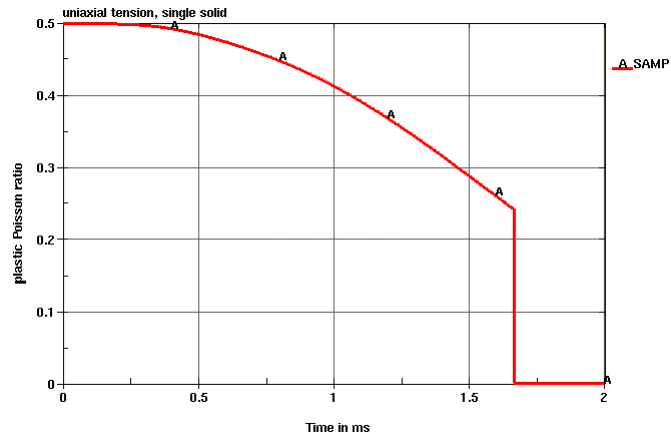
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simulation results for uniaxial tension on a single solid



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thermoplastics

bending strength of a plastic panel is higher than predicted by elasto-plastic properties derived from uniaxial tensile test

due to high compressive strength :

$$\sigma_c(\lambda) > \sigma_t(\lambda)$$

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thermoplastics

decreasing unloading modulus with increasing equivalent plastic strain
a load curve giving damage evolution for uniaxial tensile conditions can be derived from testing with unloading :

$$d(\varepsilon_p) = 1 - \frac{E(\varepsilon_p)}{E}$$

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thermoplastics

white zones indicate craze deformations in zones of localisation (caused by reduced biaxial strength)

crazing corresponds to permanent positive volumetric strain

$$v_p(\lambda) < \frac{1}{2} \quad \sigma_b(\lambda) < \sigma_t(\lambda)$$

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thermoplastics

plastic strain at failure depends upon strain rate and state of stress
separable as a first approximation

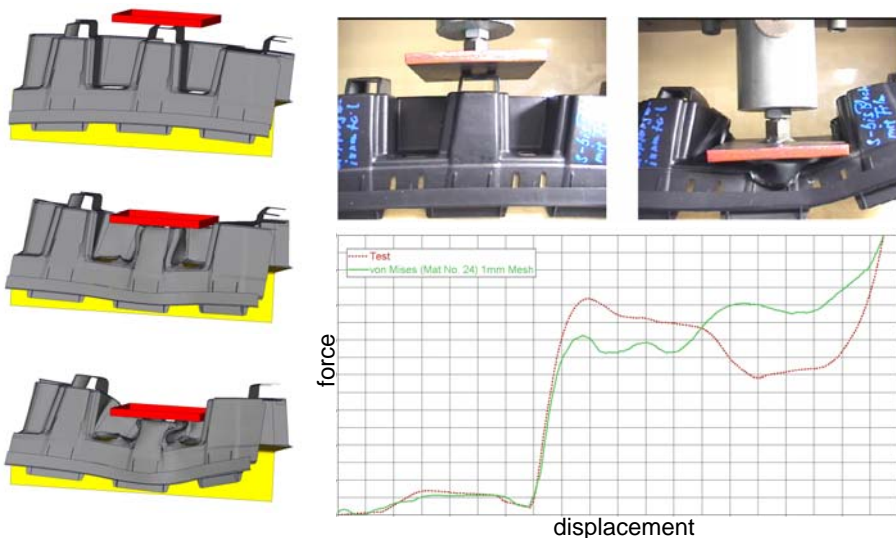
$$\varepsilon_{pf} = f(\dot{\varepsilon}_p) g\left(\frac{p}{\sigma_{vm}}\right)$$

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Validation of a Component Test (PP-T10)



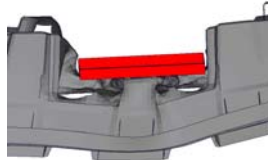
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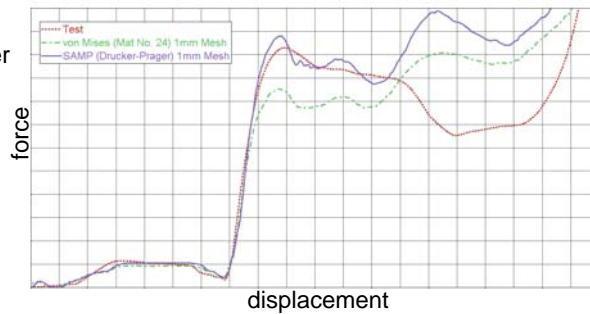
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Validation of a Component Test (PP-T10)

Typical behaviour for thermoplastics: material cards that are fitted for uniaxial tension yield a too soft responds under bending and compression

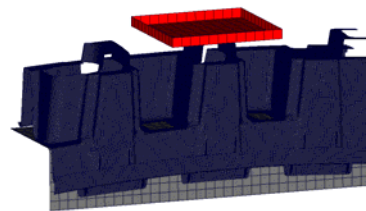
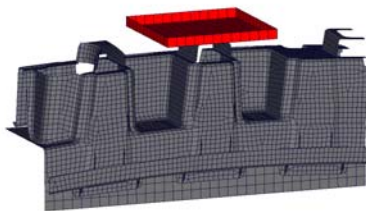


different yield curves under compression and tension necessary



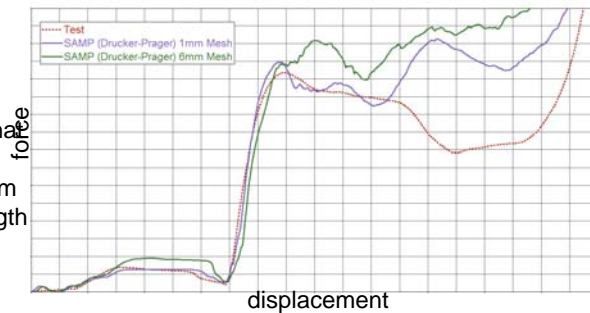
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Validation of a Component Test (PP-T10)



6mm mesh leads to stiffer response

further study has shown that mesh convergence will be reached for approx 2mm element characteristic length



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Simulation of Crazing

plastic Poisson's ratio decreases with increasing plastic strain

plastic incompressibility under compression

reduced biaxial strength

damage evolution

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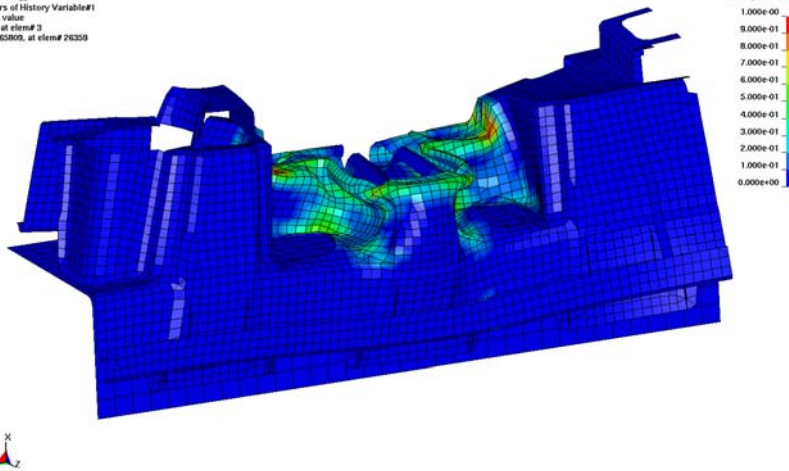
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Comparison with and without volumetric damage

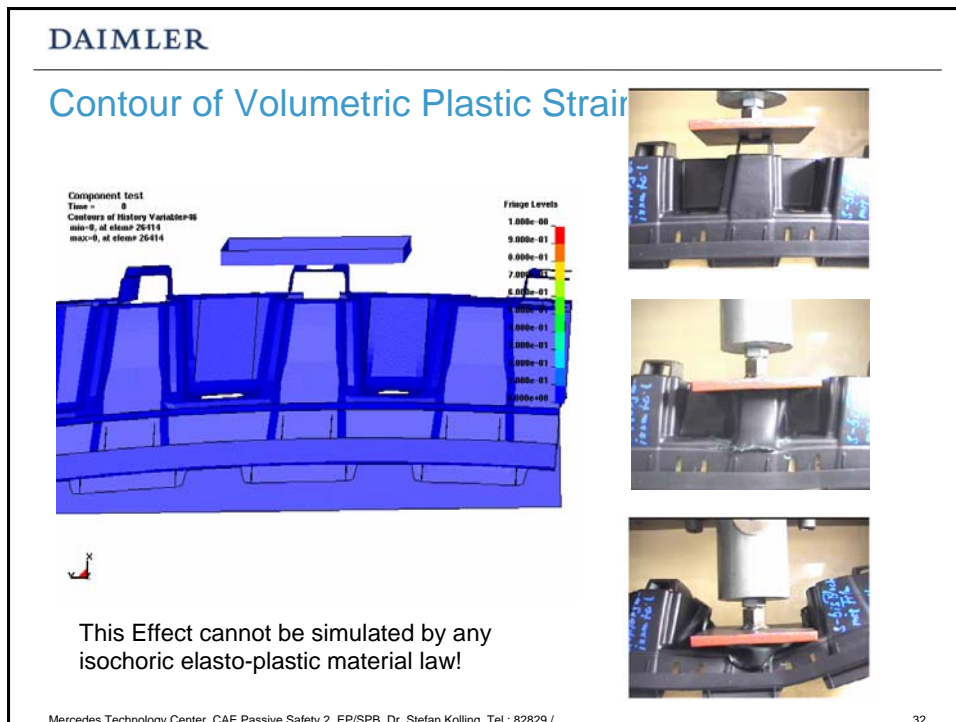
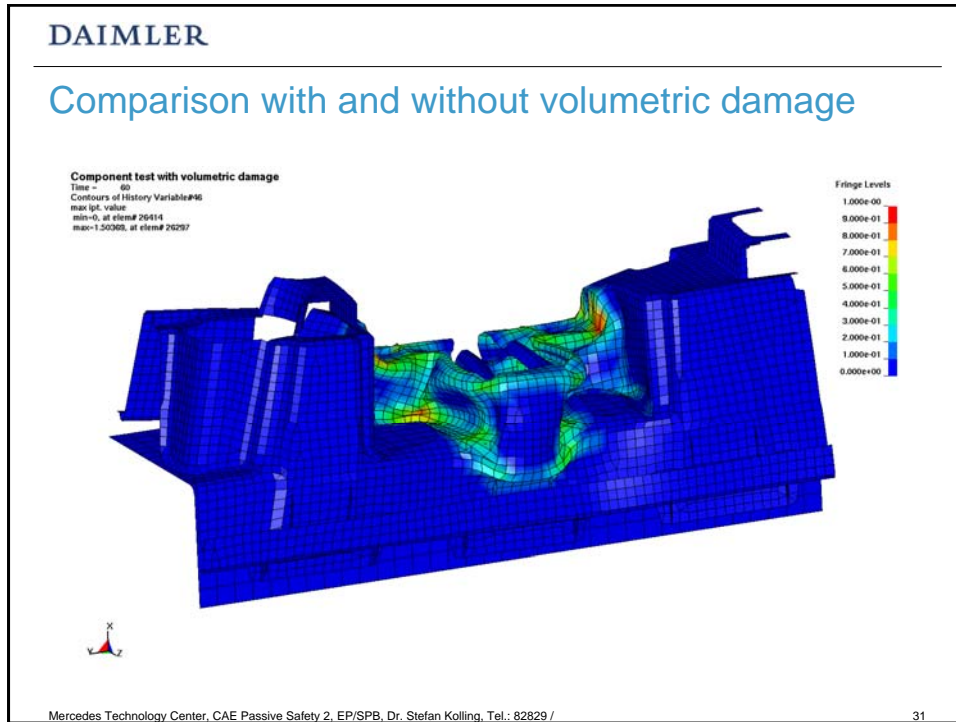
Component test without volumetric damage

Time = 40
Contours of History Variable#1
max/pl. value
min=0, at elem# 3
max=1.62800, at elem# 21259



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DAIMLER**conclusions**

matching a measured force-displacement curve in a simulation should be phase 2 of the validation process

the numerical model will have a very limited range of validity unless the deformation (and failure) mode in the simulation correspond to what was observed

it seems useful to consider a coupling between damage and volumetric plastic strain in the simulation of thermoplastics

a combination of volumetric plastic strain, damage and reduced biaxial strength allowed to simulate the craze deformation in a rather complex structural part

however the current mesh size of 6 mm is definitely too coarse for the problem under consideration